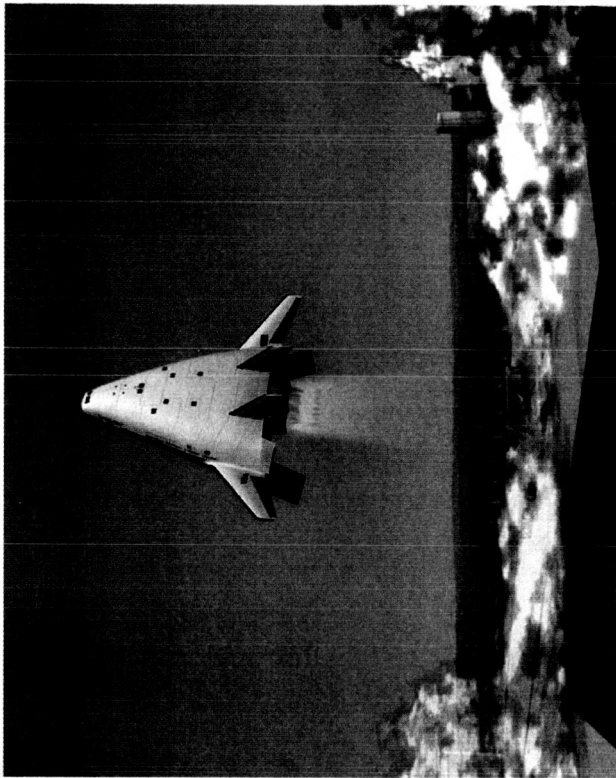
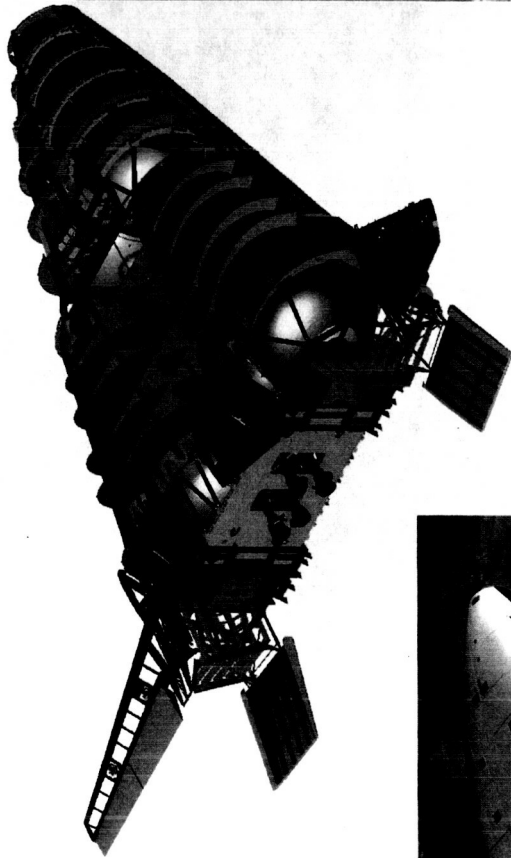
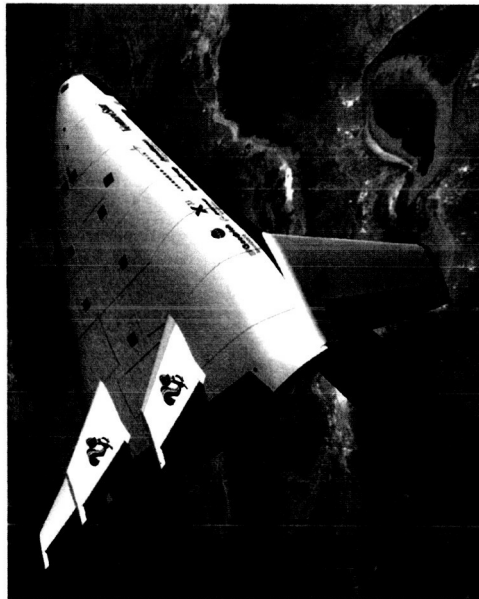


CONF PAPER/IN/24

# X-33 LH<sub>2</sub> Tank Failure Investigation Findings



*Presented by*  
Mindy Niedermeyer  
MSFC



*Aerospace Materials, Processes, and Environmental  
Technology Conference*

9/29/00

September 18 - 20, 2000

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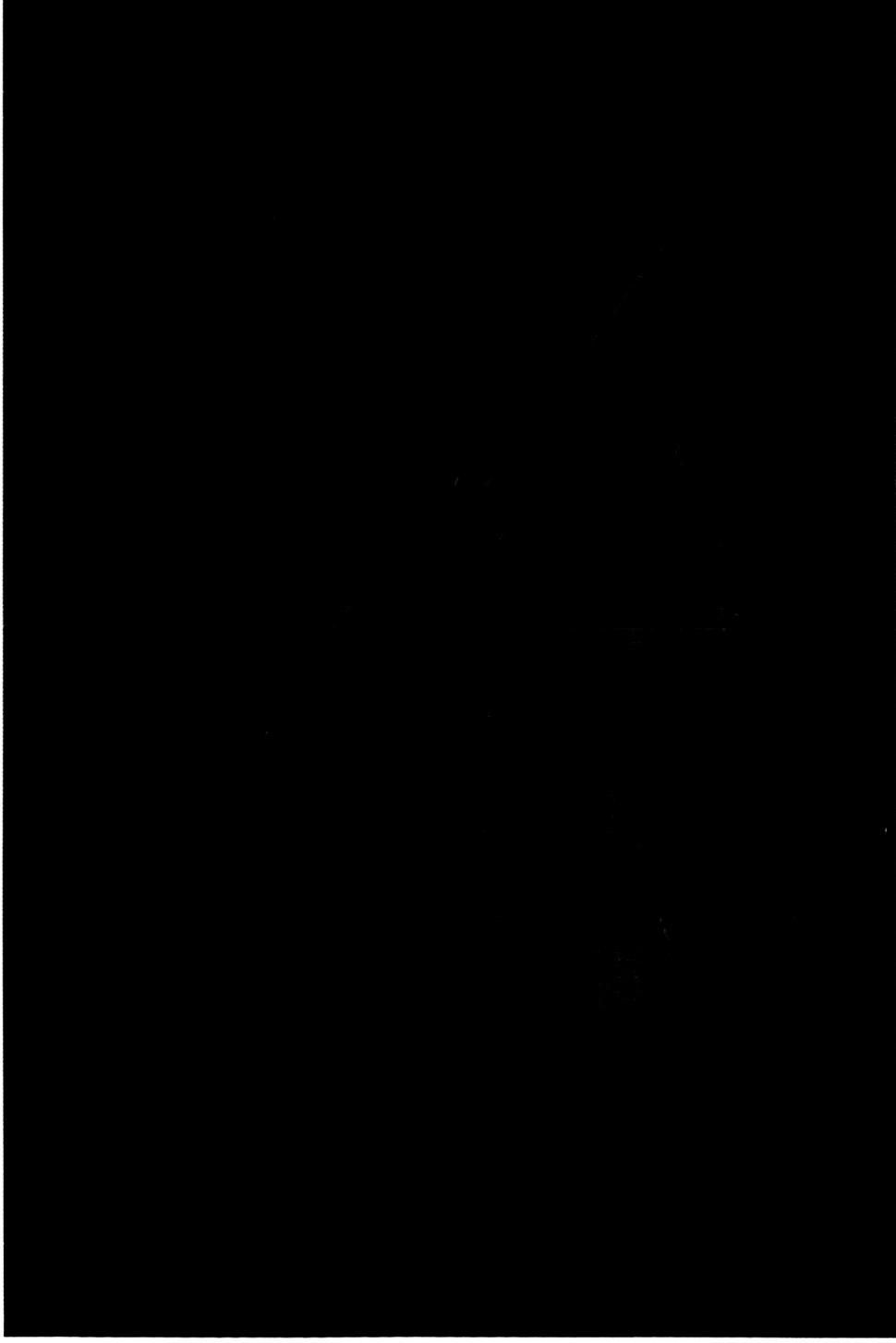
# *X-33 LH<sub>2</sub> Tank Failure Investigation Findings*



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## ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

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# ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

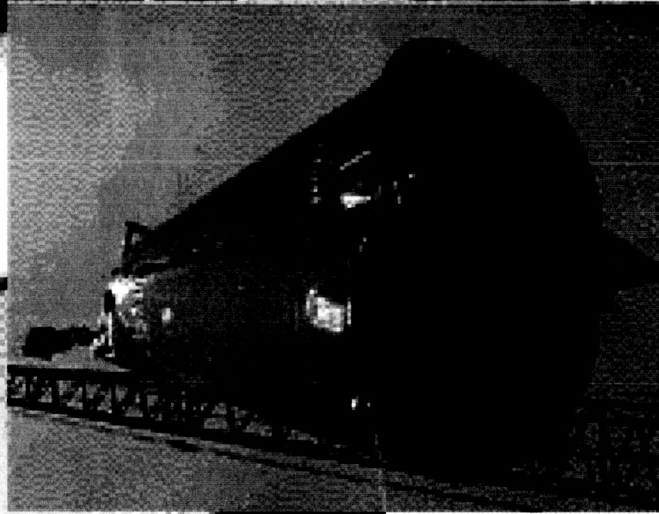
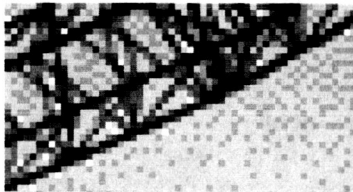
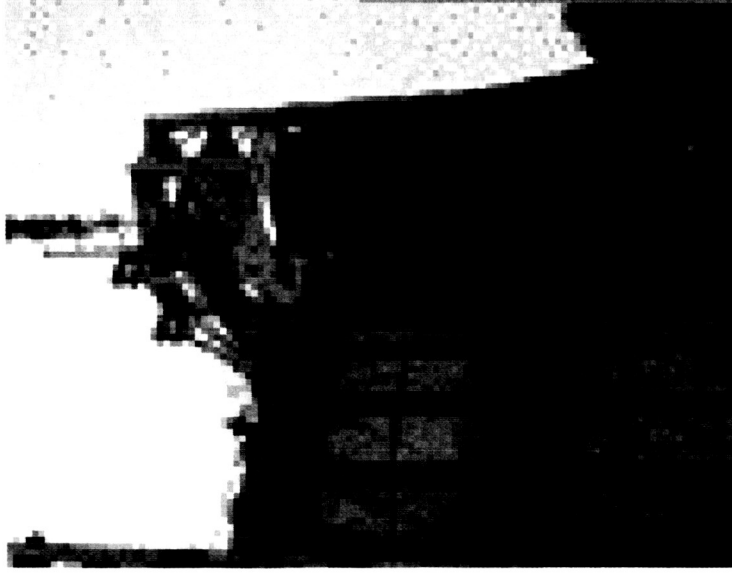
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## **Introduction**

- Tank History and Test Objectives**
- Failure Description**
- Investigation**
- Conclusions**



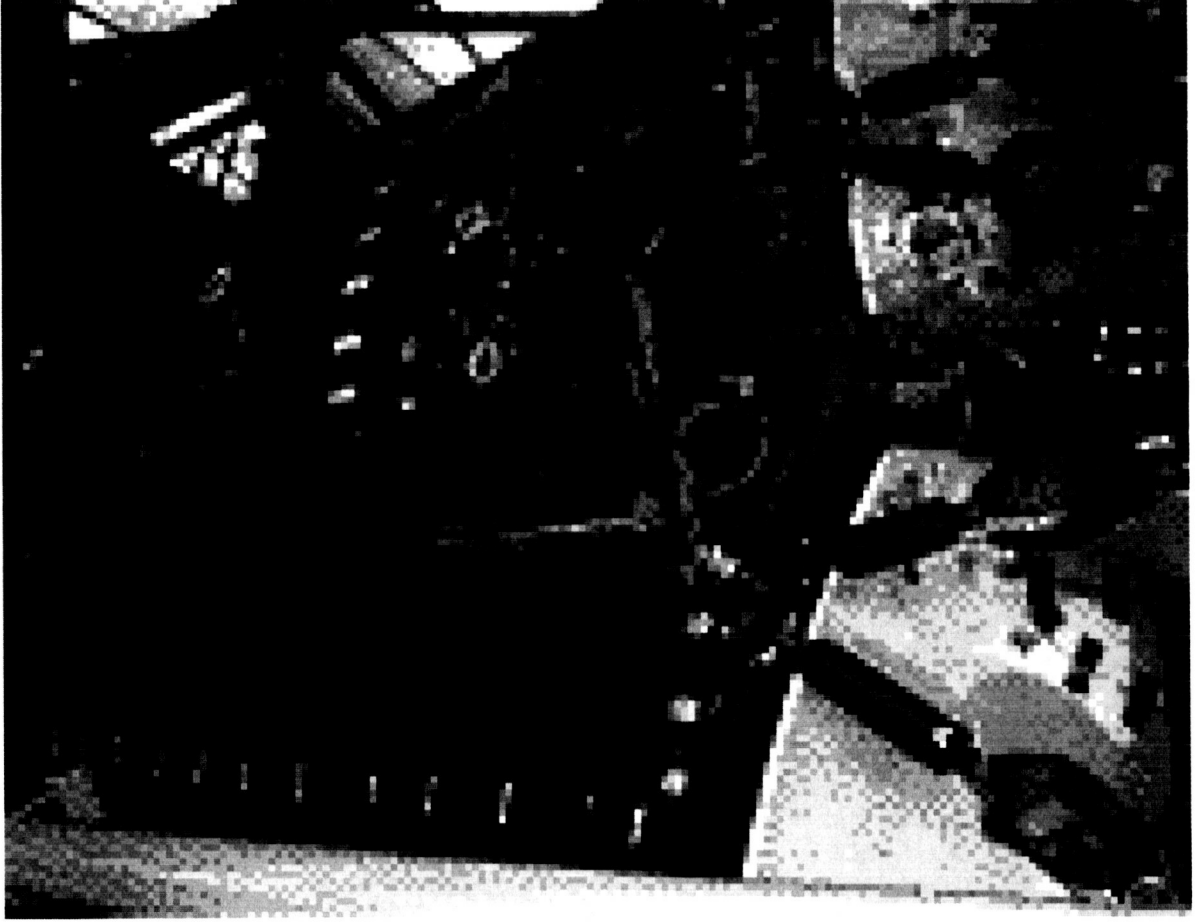
# *X-33 LH<sub>2</sub> Tank Failure Investigation Findings*



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## ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

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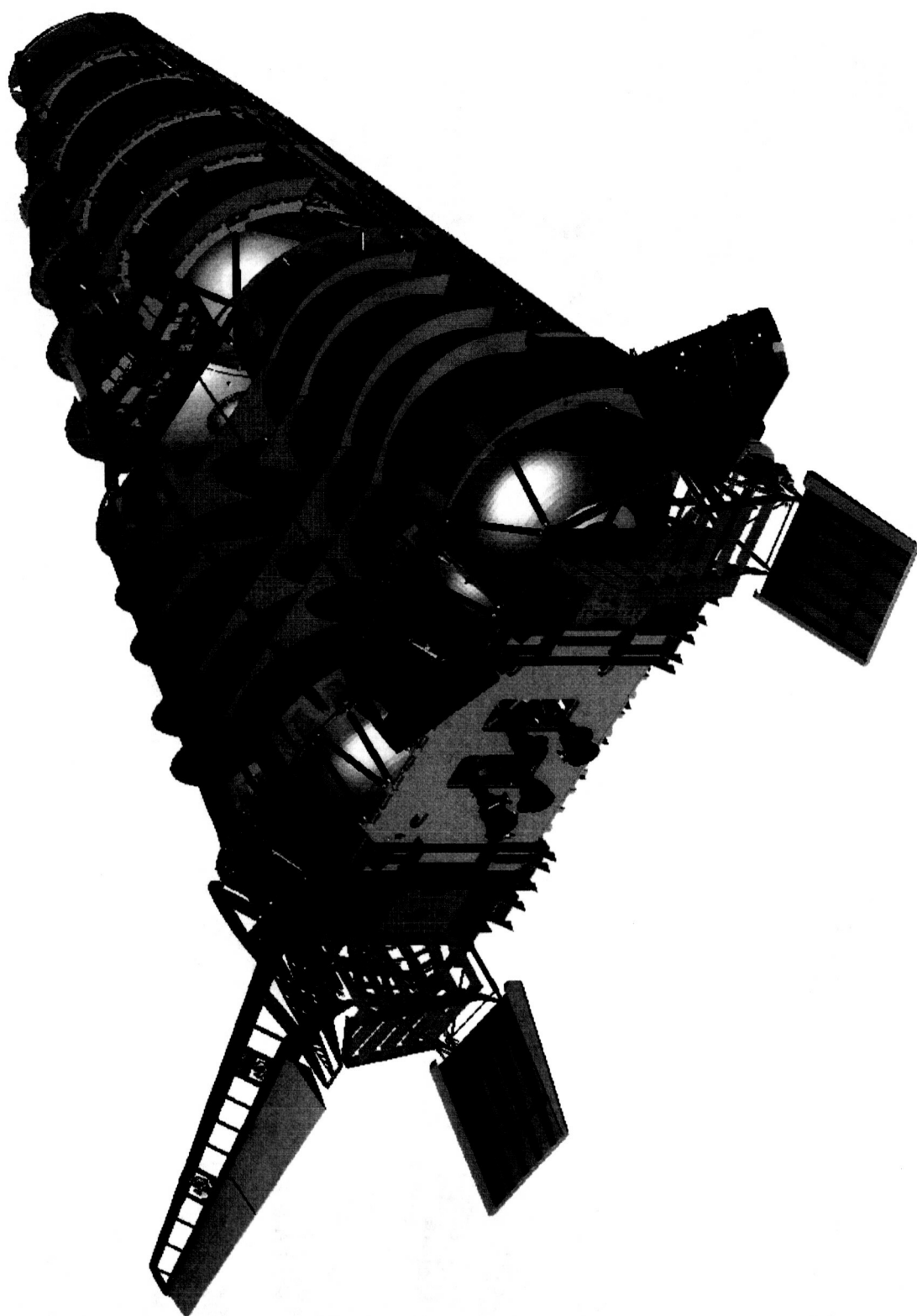
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# ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

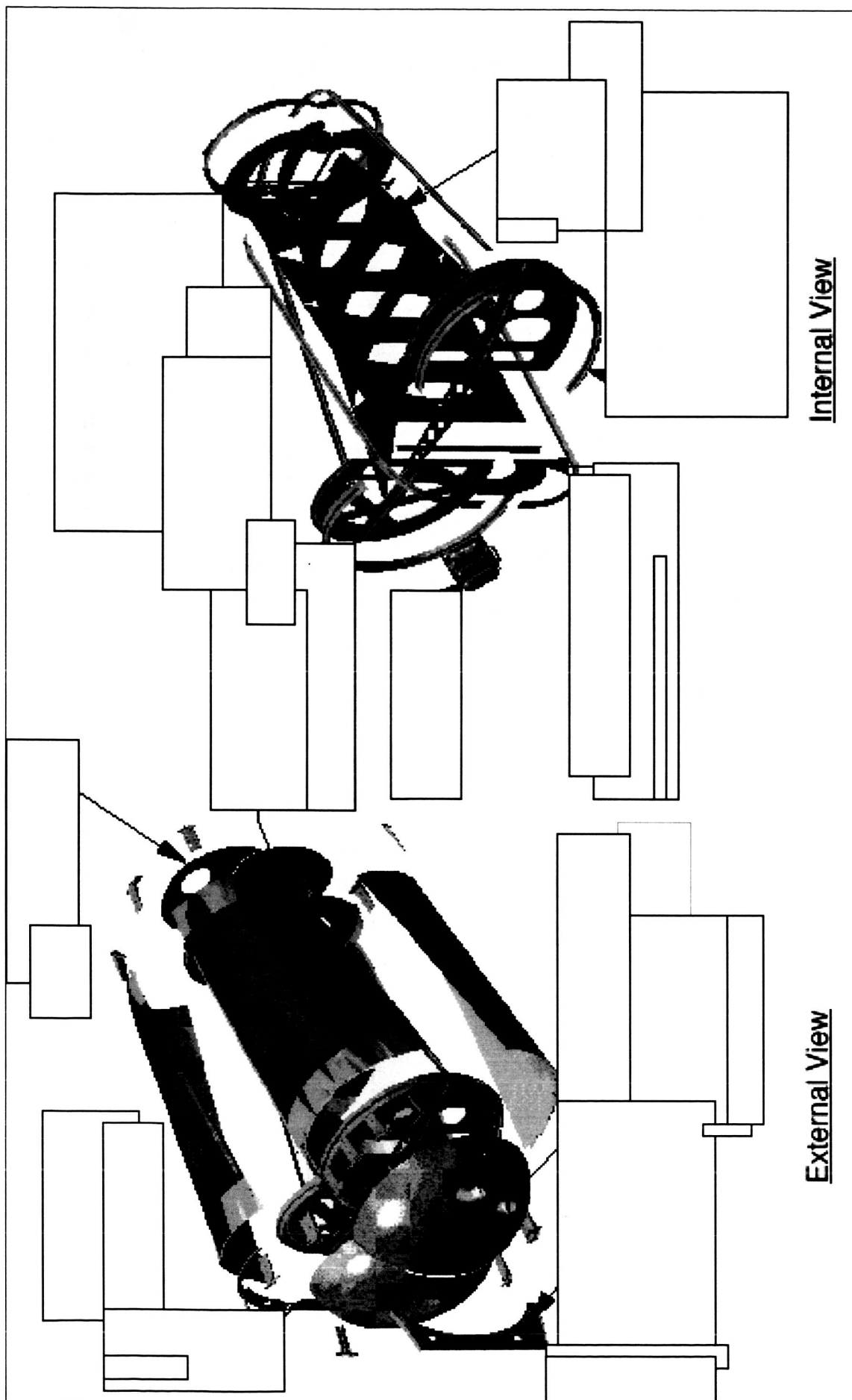
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## **Tank Description**

- **Structural component of the aft body**
- **Quad-lobe design**
- **Sandwich - honeycomb graphite epoxy construction**



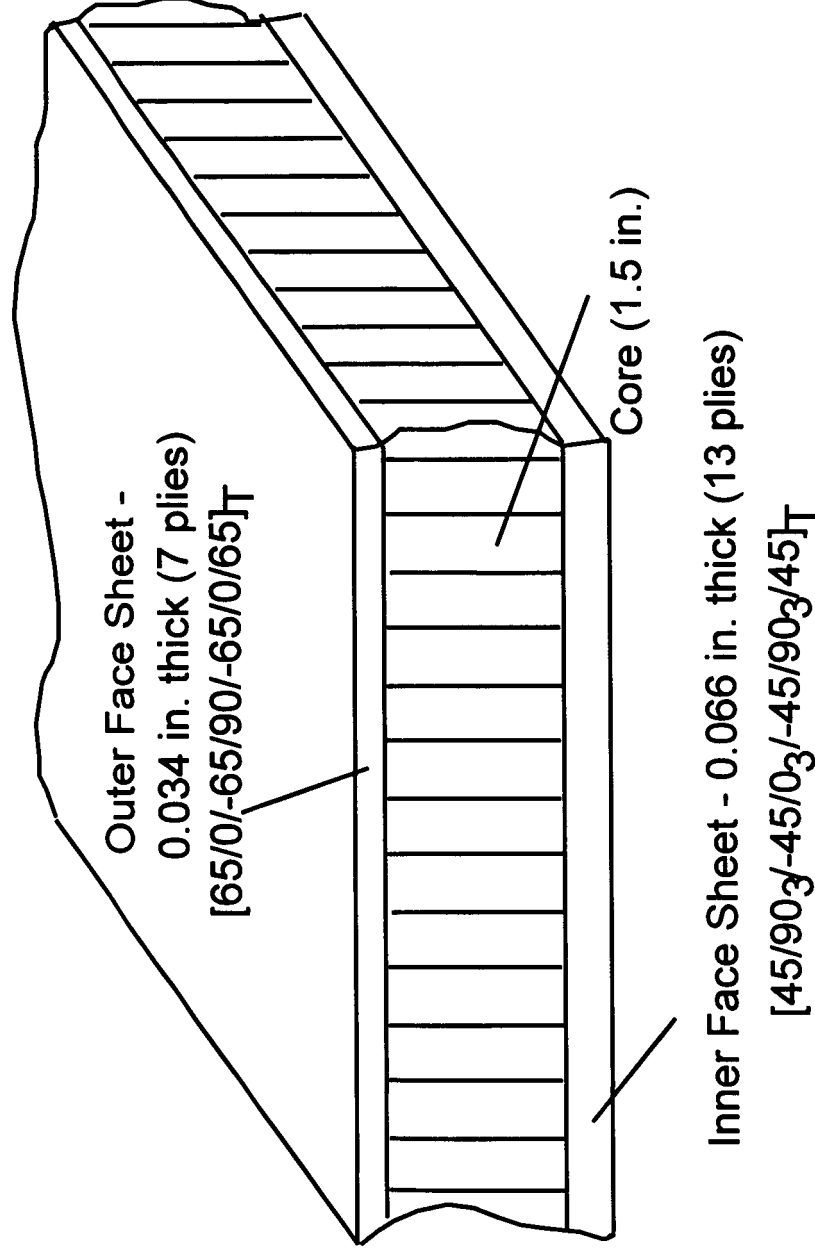
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External View

Internal View

# *X-33 LH<sub>2</sub> Tank Failure Investigation Findings*



## **Geometry of sandwich structure**

# ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

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## **Test Objectives**

- Verify structural integrity at 105% expected flight load limit varying the following parameters**
- Cryogenic temperature**
- Internal pressure**
- Mechanical loading**

## ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

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- **September 21, 1999 test aborted due to hydrogen leaks**
  - **100% cryogen fill (LH2)**
  - **20 psig internal pressure**
- **November 3, 1999 test completed**
  - **100% cryogen fill (LH2) at 42 psig internal pressure**
  - **Load case 5 applied at 5 psig internal pressure**
  - **Tank drained of cryogen**

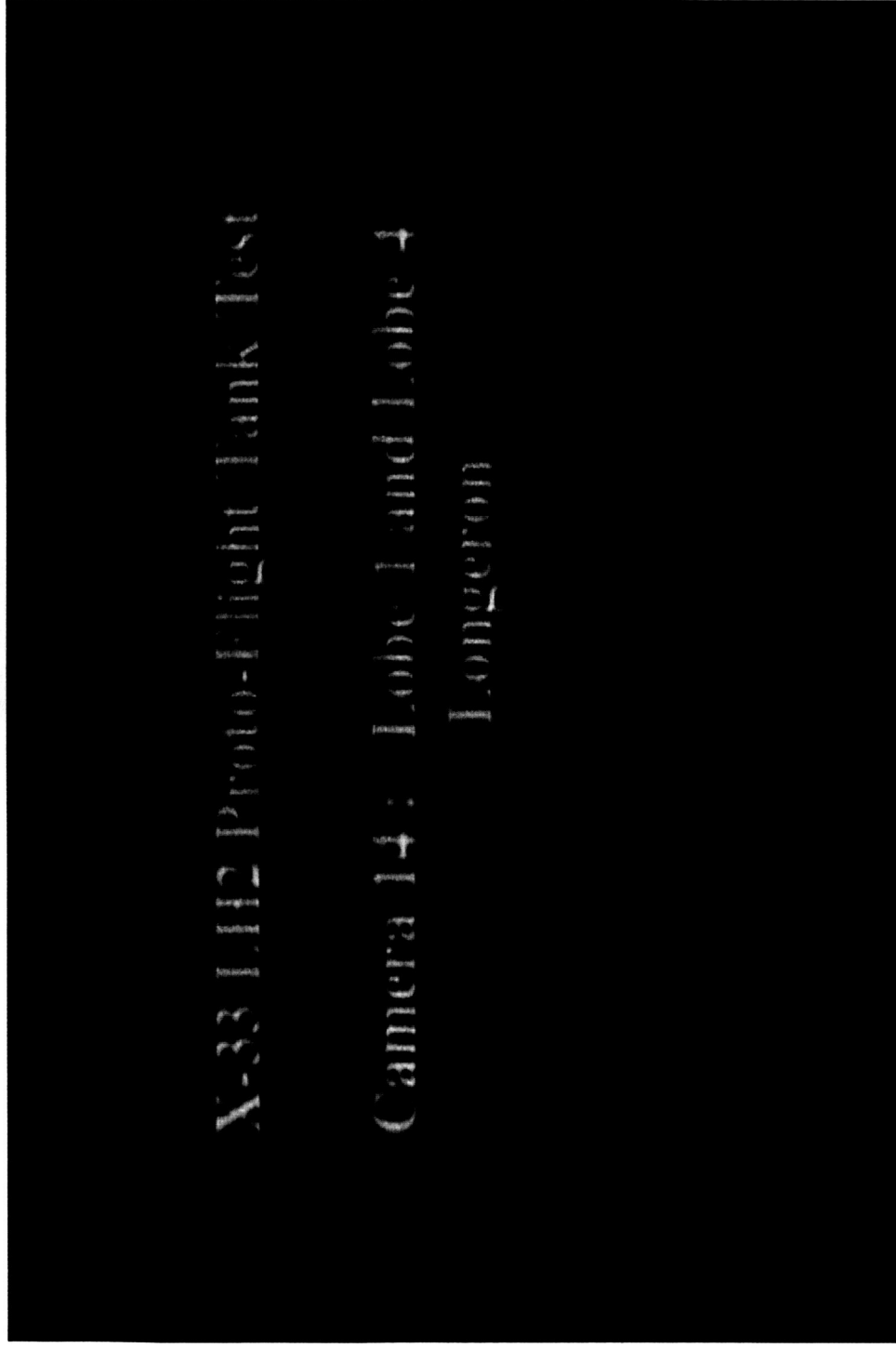


## ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

### **Timeline**

- **Tank filled, 12:30 PM**
- **Tank pressurized to 42 psig, 2:00 PM**
- **Tank vented to 2 psig, 3:00 PM**
- **Loads applied, pressure increased to 5 psig, 4:40 PM**
- **Tank drained, 6:00 PM**
- **Lobe 1 failure, 6:24 PM**

## X-33 LH<sub>2</sub> Tank Failure Investigation Findings

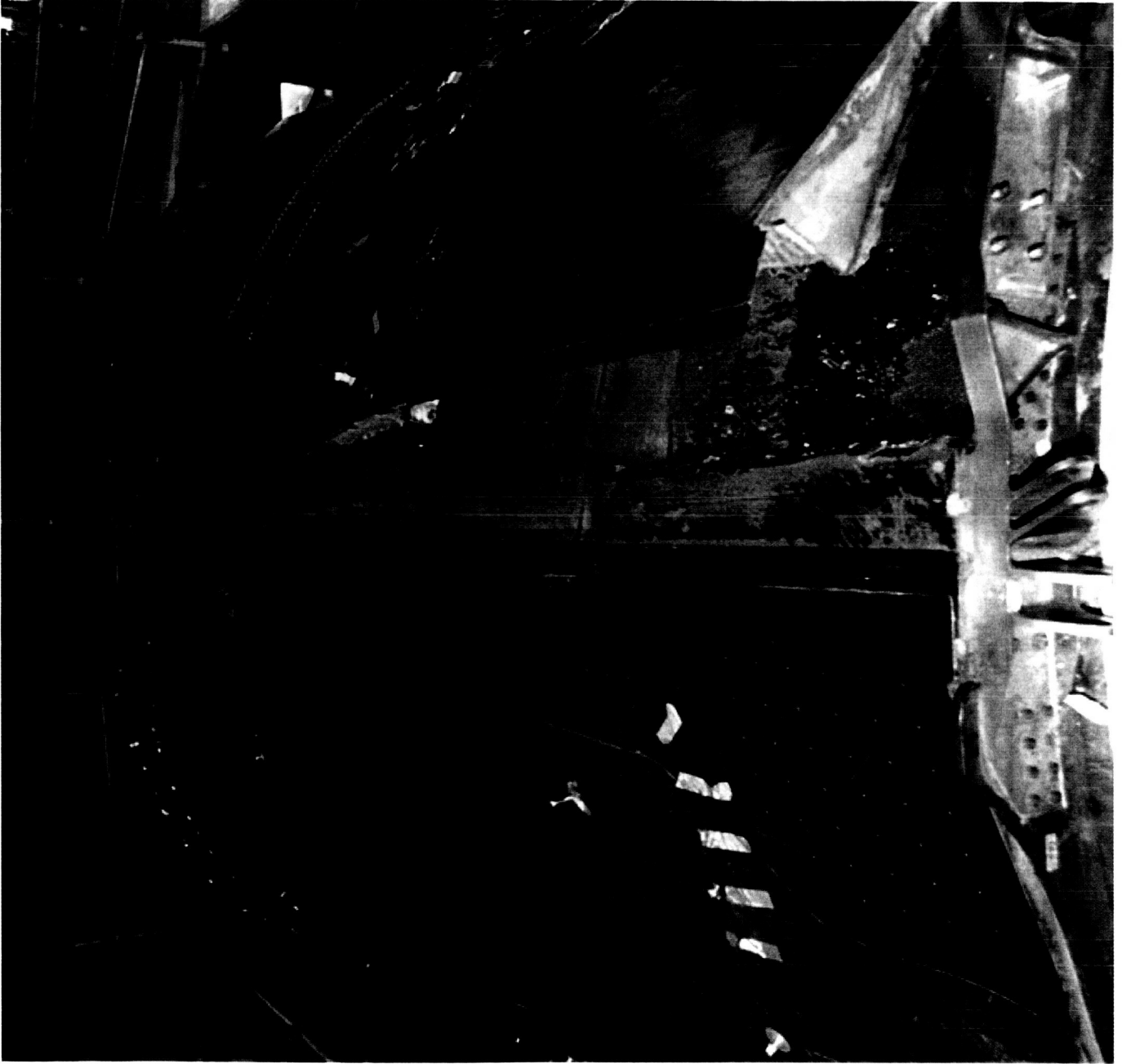


# ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

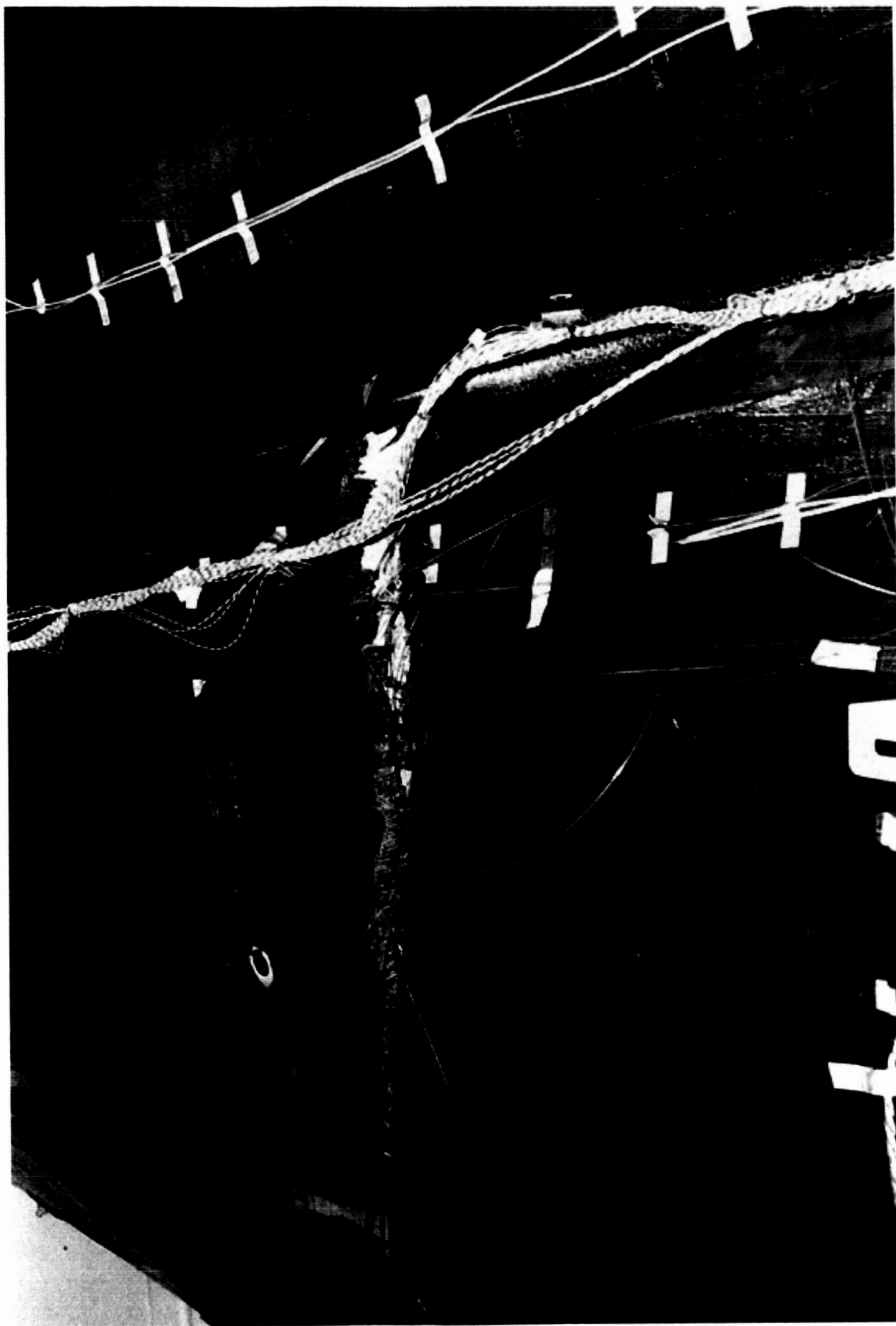
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## **Initial Findings**

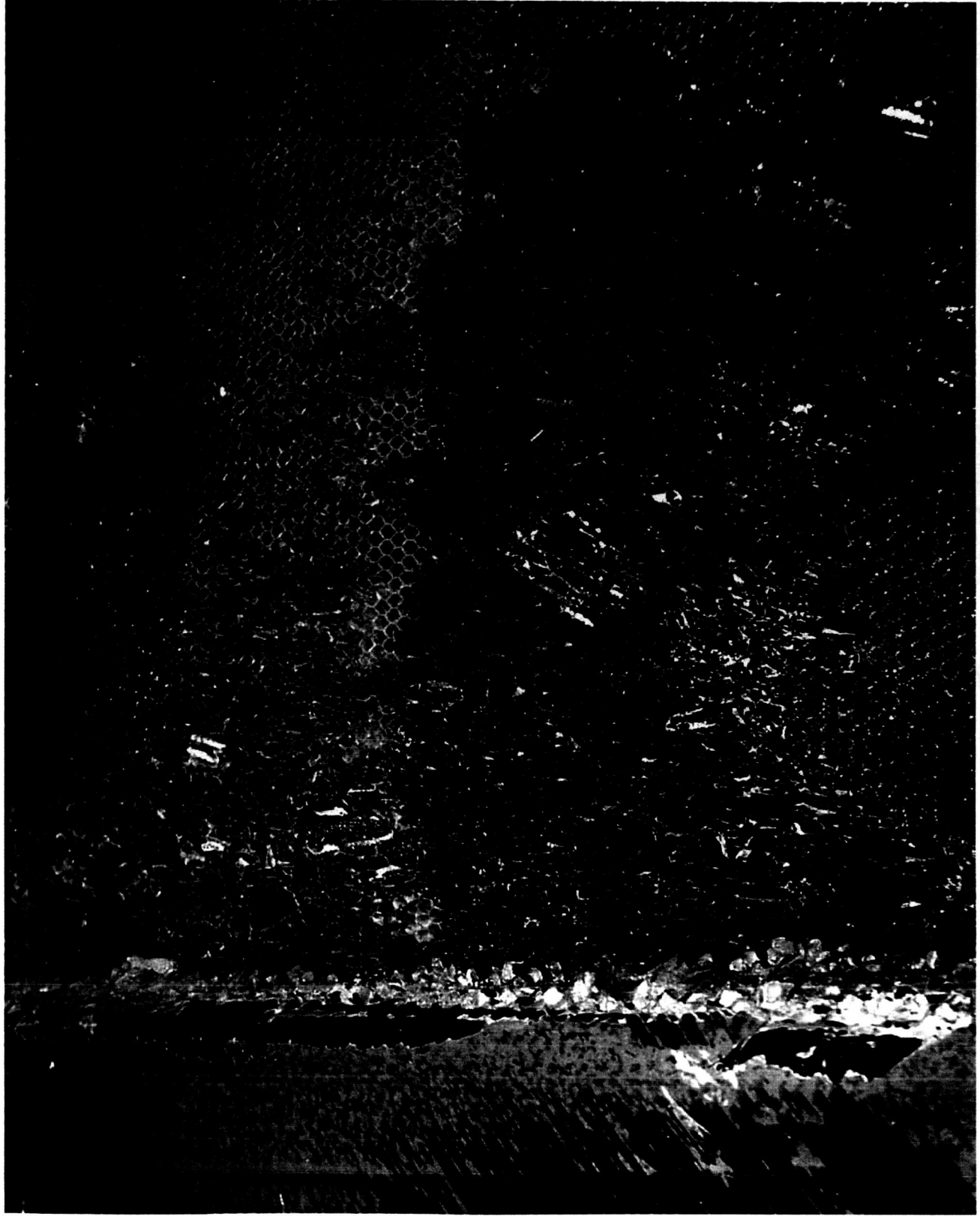
- **Peel Failure**
  - **Outer skin and core peeled away from inner skin**
- **Core Failure**
  - **Core is 'mangled'**
- **Hydraulic fluid on test article**
- **Foreign Object Debris (FOD)**
- **Poor bondlines**
- **Pressure in core above ambient 13 hours after failure**



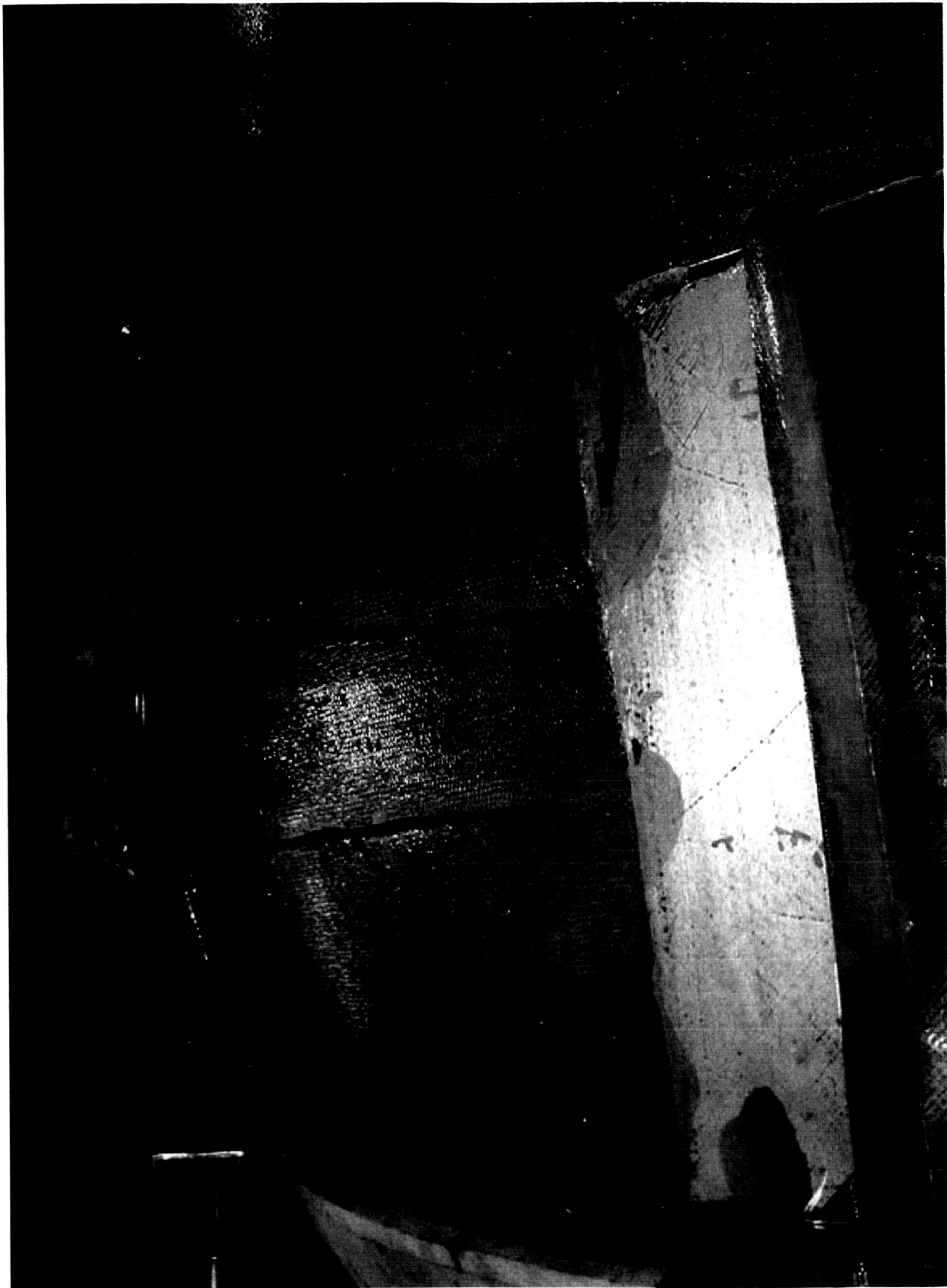
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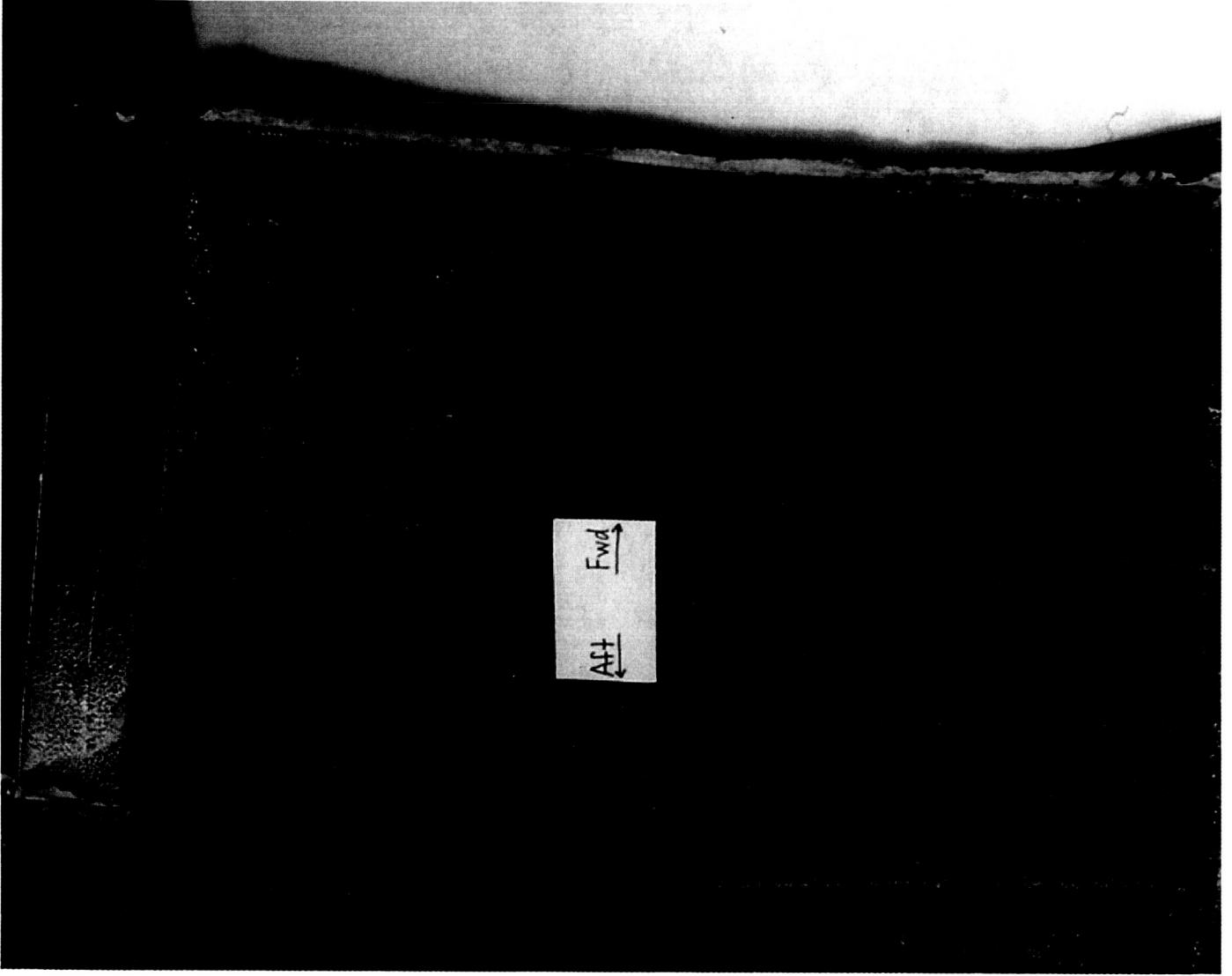
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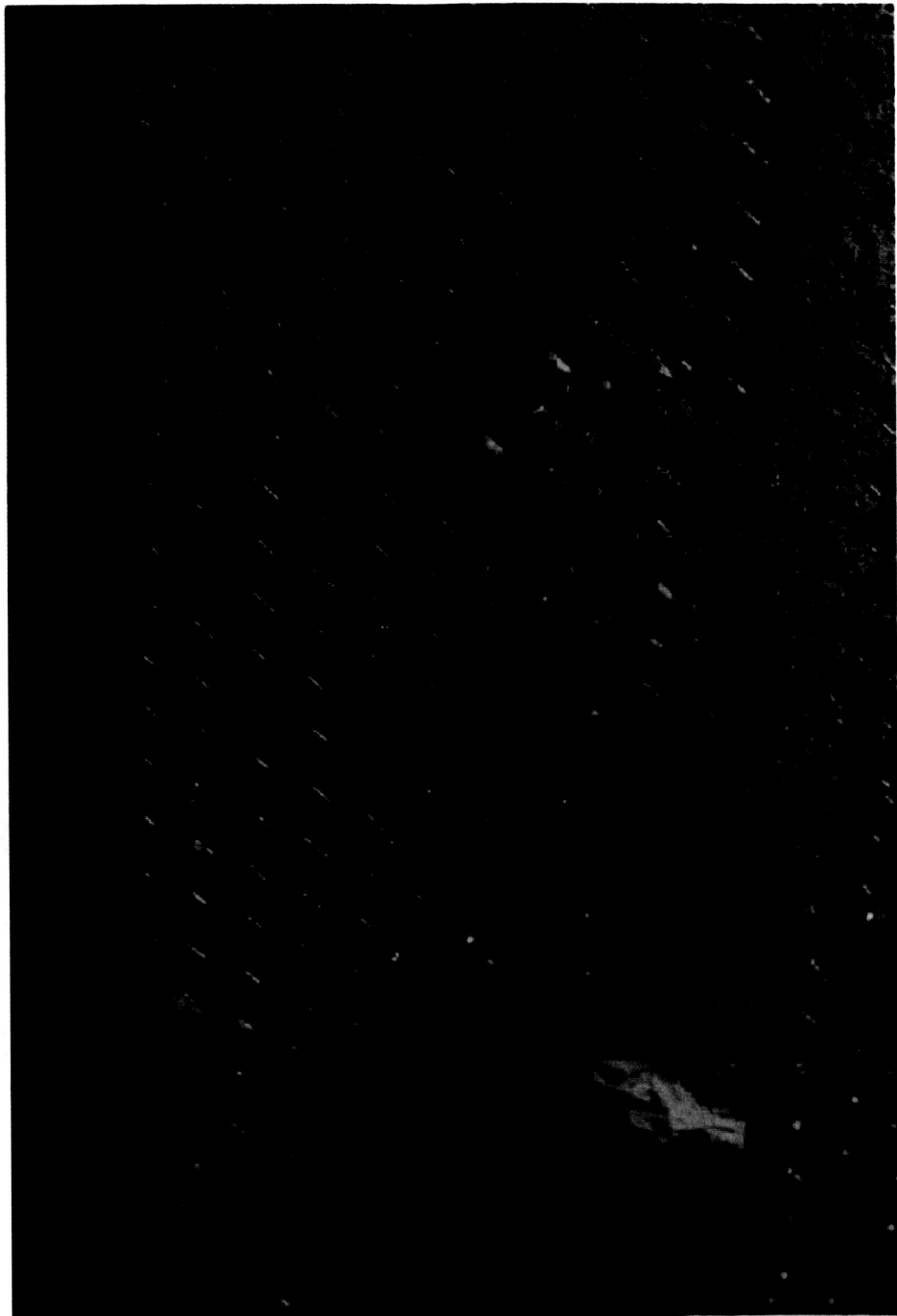




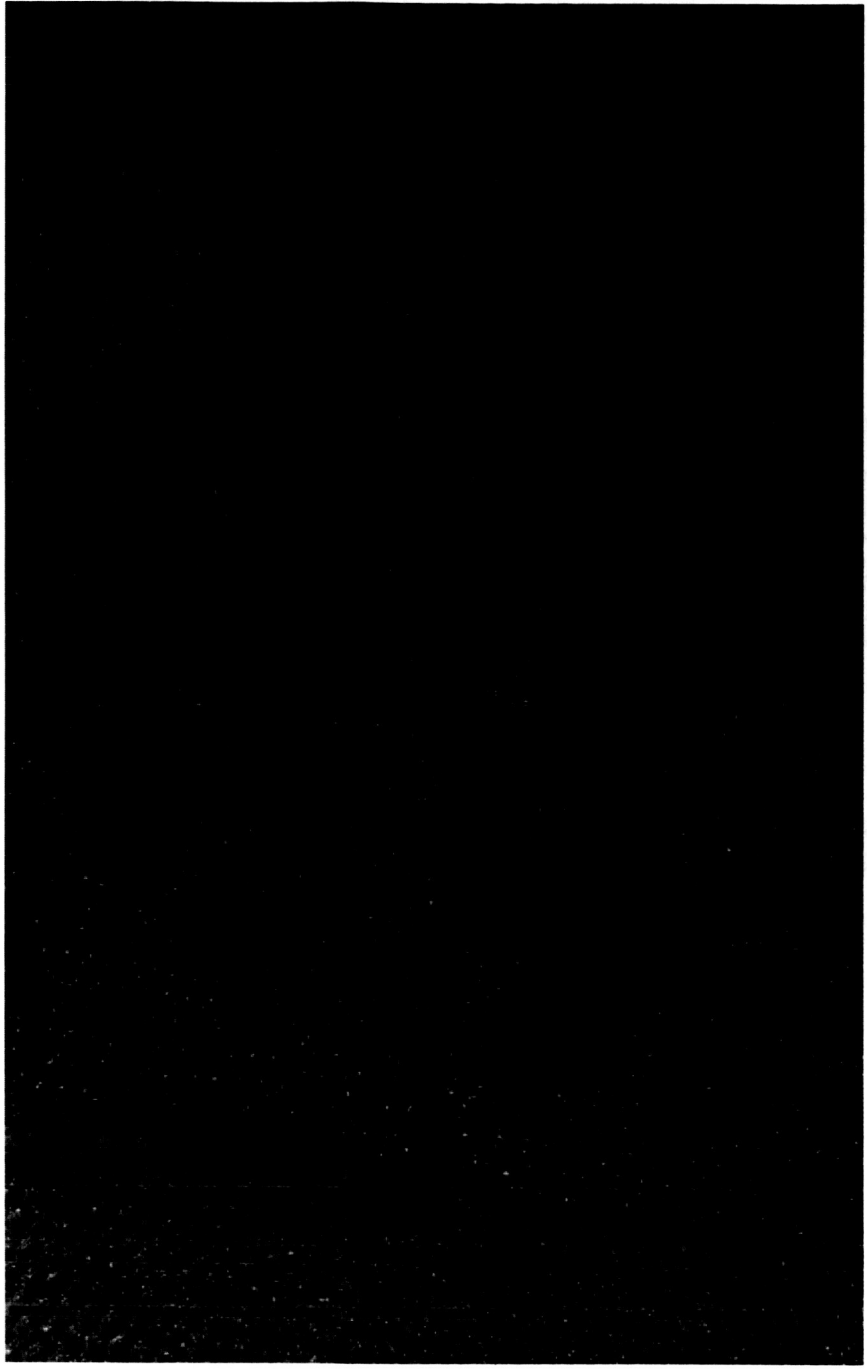
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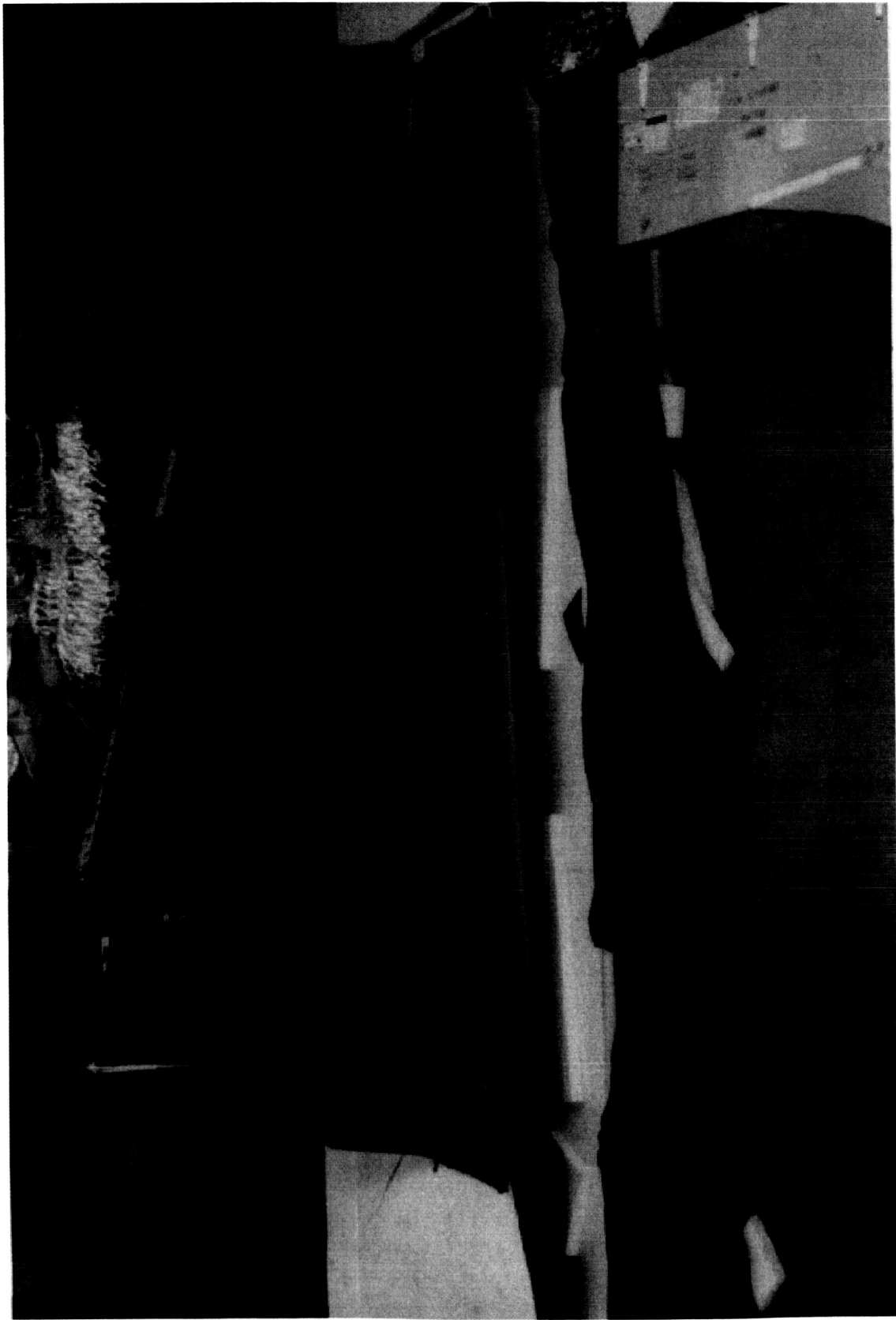
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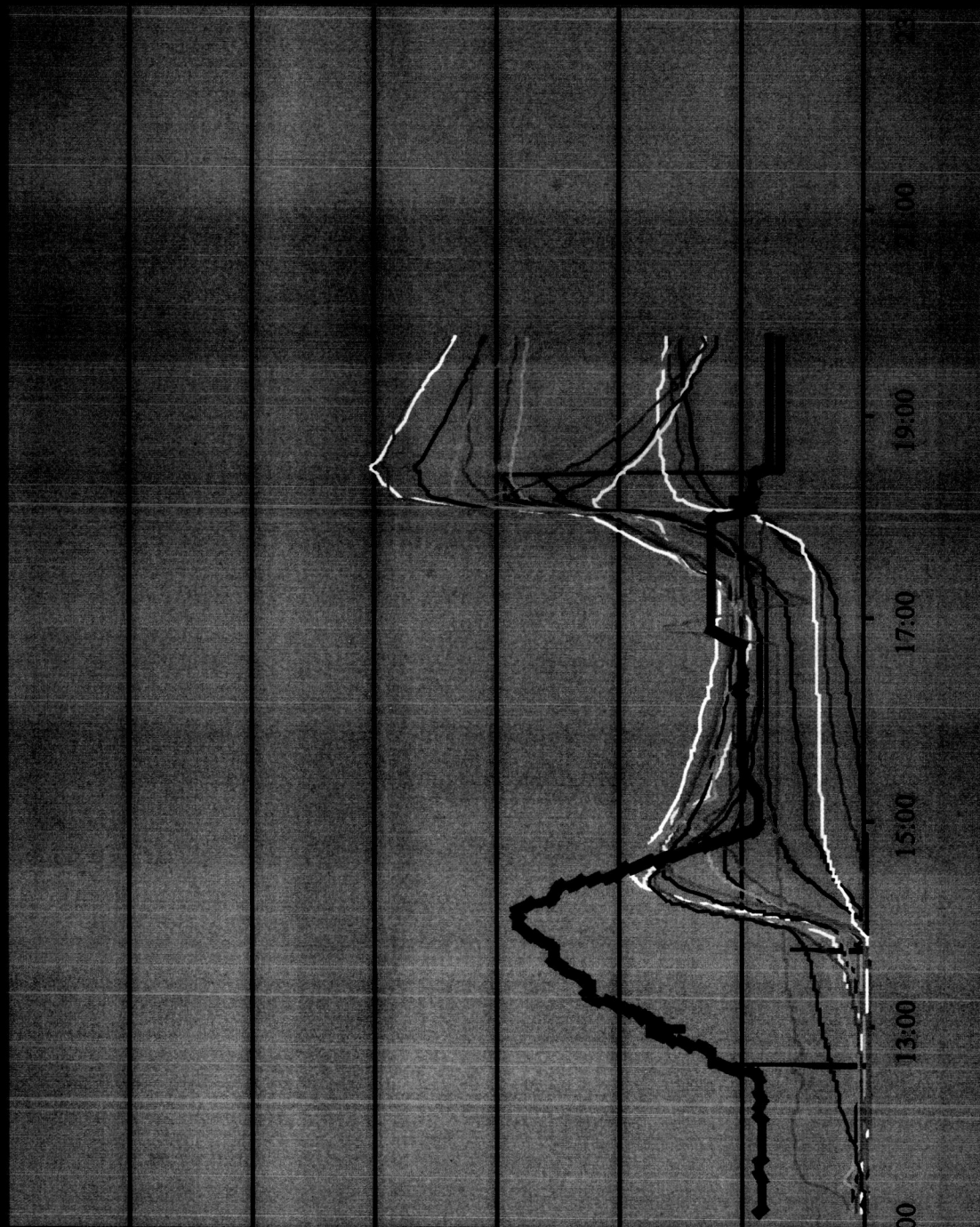
# ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

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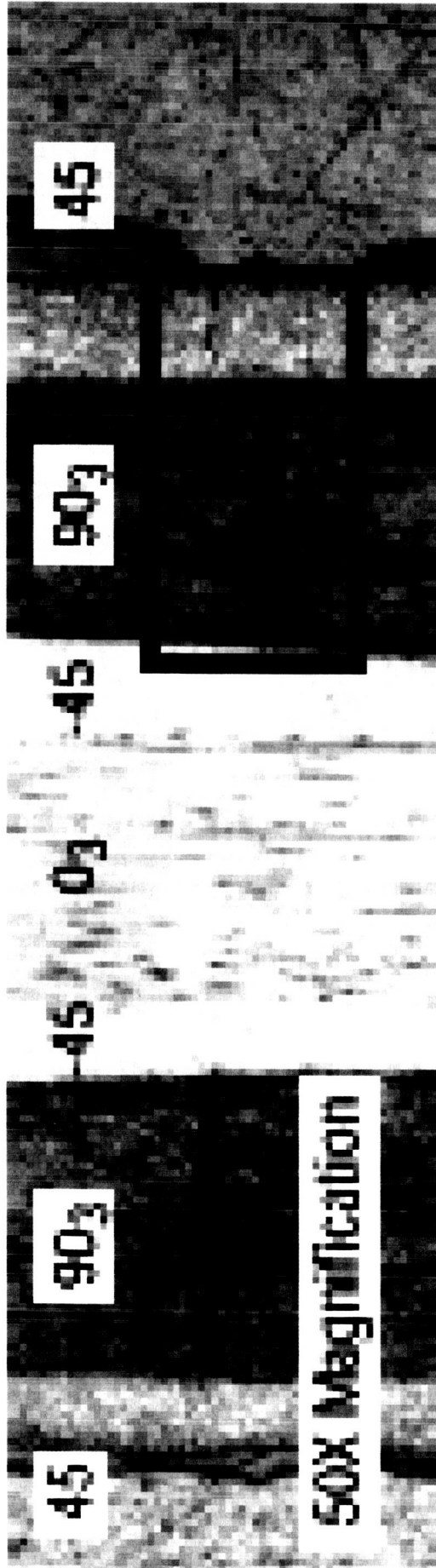
## **Subsequent Findings**

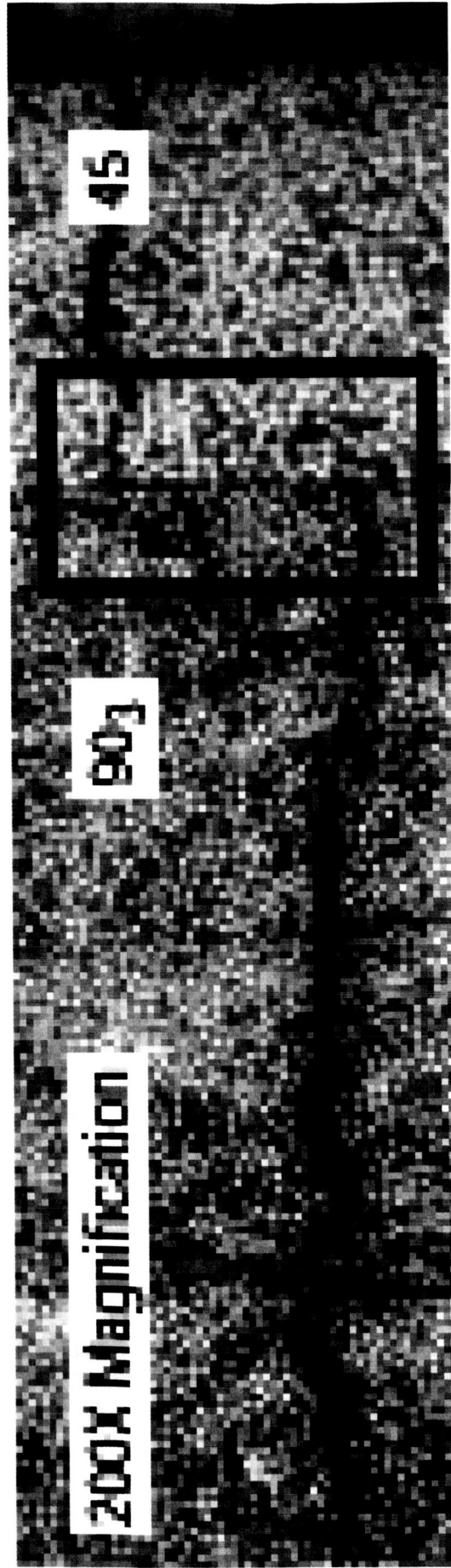
- Core pressures behaved unexpectedly
- Microcracking

# re Pressure vs T



Pressure (psia)







500X Magnification

90%

45

# ***X-33 LH<sub>2</sub> Tank Failure Investigation Findings***

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## **Conclusions**

- The inner skin microcracked and hydrogen infiltrated
- The cracks grew larger under pressure
- When pressure was removed cracks closed slightly
- When tank was drained and warmed, cracks closed and blocked leak path
- FOD and debond areas provided an opportunity for a leak path
- There is still hydrogen in the other 3 lobes today